



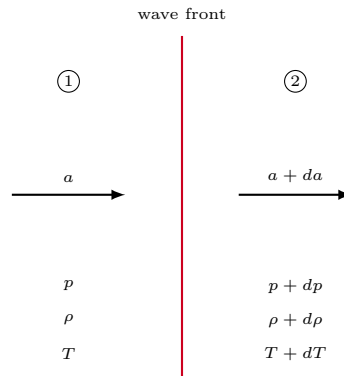
# Compressible Flow TME085

One-Dimensional Steady Flow

Speed of Sound

Division of Fluid Dynamics  
Department of Mechanics and Maritime Sciences  
Chalmers University of Technology

# Sound Wave



Figur 1: sound wave

In Fig. 1, station 1 represents the flow state ahead of the sound wave and station 2 the flow state behind the sound wave. Set up the continuity equation for one-dimensional flows between 1 and 2. If we could change frame of reference and follow the sound wave, we would see fluid approaching the wave with the propagation speed of the wave,  $a$ , and behind the wave, the fluid would have a slightly modified speed,  $a + da$ . There would also be a slight in all other flow properties. Let's apply the one-dimensional continuity equation between station 1 and station 2.

$$\rho_1 u_1 = \rho_2 u_2$$

$$\rho a = (\rho + d\rho)(a + da)$$

$$\rho a = \rho a + \rho da + a d\rho + \underbrace{d\rho da}_{\sim 0} \Rightarrow$$

$$a = -\rho \frac{da}{d\rho} \tag{1}$$

The one-dimensional momentum equation between station 1 and station 2 gives

$$\rho_1 u_1^2 + p_1 = \rho_2 u_2^2 + p_2$$

$$\rho a^2 + p = (\rho + d\rho)(a + da)^2 + (p + dp)$$

$$\rho a^2 + p = \rho a^2 + 2\rho a da + \underbrace{\rho da^2}_{\sim 0} + d\rho a^2 + \underbrace{2d\rho a da}_{\sim 0} + \underbrace{d\rho da^2}_{\sim 0} + p + dp \Rightarrow$$

$$dp = -2\rho a da - d\rho a^2 \Rightarrow$$

$$da = -\frac{dp + d\rho a^2}{2\rho a} = -\frac{d\rho}{2a\rho} \left( \frac{dp}{d\rho} + a^2 \right) \Rightarrow$$

$$\frac{da}{d\rho} = -\frac{1}{2a\rho} \left( \frac{dp}{d\rho} + a^2 \right) \quad (2)$$

Eqn. 2 in 1 gives

$$a = \frac{1}{2a} \left( \frac{dp}{d\rho} + a^2 \right) \Rightarrow$$

$$a^2 = \frac{dp}{d\rho} \quad (3)$$

Sound wave:

- gradients are small
- irreversible (dissipative effects are negligible)
- no heat addition

Thus, the change of flow properties as the sound wave passes can be assumed to be an isentropic process

$$a^2 = \left( \frac{dp}{d\rho} \right)_s \quad (4)$$

$$a = \sqrt{\left(\frac{dp}{d\rho}\right)_s} = \sqrt{\frac{1}{\rho\tau_s}} \quad (5)$$

where  $\tau_s$  is the compressibility of the gas. Eqn. 5 is valid for all gases. It can be seen from the equation, that truly incompressible flow ( $\tau_s = 0$ ) would imply infinite speed of sound.

Since the process is isentropic, we can use the isentropic relations if we also assume the gas to be calorically perfect

$$\frac{p_2}{p_1} = \left(\frac{\rho_2}{\rho_1}\right)^\gamma \Rightarrow p = C\rho^\gamma$$

$$a^2 = \left(\frac{dp}{d\rho}\right)_s = \gamma C\rho^{\gamma-1} = \gamma \underbrace{[C\rho^\gamma]}_{=p} \rho^{-1} = \frac{\gamma p}{\rho} \Rightarrow$$

$$a = \sqrt{\frac{\gamma p}{\rho}} \quad (6)$$

or

$$a = \sqrt{\gamma RT} \quad (7)$$